# Boundary element analysis of elastic fields in non-horizontally layered halfspace whose horizontal boundary subject to tractions 

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#### Abstract

This paper presents a boundary element analysis of linear elastic fields in a layered halfspace whose material interface planes are not parallel to its horizontal boundary surface. This boundary element analysis uses the generalized Kelvin solution in a multilayered elastic solid (or the so-called Yue's solution) for taking into account the non-horizontally layered structures. It also adopts the infinite boundary elements for evaluating the influence of a far-field region. It further adopts both the discontinuous finite and infinite boundary elements to discretize the boundary surface around the strike lines of the inclined material interfaces. It uses the Kutt's numerical quadrature to evaluate the strongly singular integral in the discretized boundary integral equation. Numerical examples are presented to illustrate the effects of the non-horizontally layered materials to the displacements and stresses induced by the tractions on the horizontal boundary surface. Two non-horizontally layered halfspace models are used for numerical analysis and illustrations. Numerical results show that across the material interface, the elastic displacements are non-smoothly continuous to different degrees and some stress components can have very high values at and adjacent the interface planes, which can be important to tensile or shear failure in non-homogeneous materials.


Key words: BEM; generalized Kelvin solution; halfspace; layered solid; non-horizontal interface; infinite elements; elasticity

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## 1. Introduction

### 1.1. Horizontally layered halfspace model

Layered solid materials widely exist in nature or man-made structures. Their responses to external loadings before failure can be modelled with the responses of elastic halfspace model subject to the same external loading. The elastic halfspace model can compose of the layered elastic materials with the same mechanical properties and geometries of the actual layered solid materials. The interfaces of the layered solid materials may or may not be parallel to their boundary surfaces. Many investigations have been done for the elastic responses of the horizontally layered halfspaces subject to tractions or other types of loading conditions by many researchers since 1940s (e.g., Holl [1], Burminster [2,3,4], King [5], Singh [6] and Pan [7]). A relatively complete list of relevant publications can be found in the recent publications by Yue [8,9].

### 1.2. Non-horizontally layered halfspace model

As shown in Fig. 1, this paper examines the elastic responses of an elastic halfspace with non-horizontally layered solid materials due to the action of tractions at the horizontal boundary surface. There is a very limited investigation on the elastic responses of such non-horizontally layered halfspaces subject to tractions. Two examples of relevant studies are given by Pereira and Parreira [10] and Moser et al. [11] respectively. Due to its importance in science and engineering and its difficulties in analytical or numerical formulations, this paper aims to develop a novel boundary element method to calculate the elastic responses of the displacement and stress fields with high efficiency and accuracy.

### 1.3. Classical boundary element method and issues

The classical boundary element method (BEM) is ideally suited for the analysis of elastic solid materials occupying either full-space or halfspace because BEM only needs to discretize the external boundary surface and automatically satisfies the conditions at the infinity. The classical Kelvin's solution is commonly used as the fundamental singular solution in classical BEM. When it is applied to a layered solid as shown in Fig. 1, it needs to divide the layered
solid into many homogeneous domains. Furthermore, it needs to discretize the internal interfaces of the layered materials so that the continuity conditions at the internal interfaces can be formulated numerically together with those at the external boundary surfaces. For example, Pereira and Parreira [10] and Moser et al. [11] adopted this multi-region method of BEMs and analyzed the elastic responses of a halfspace with two non-horizontal layers subject to tractions. As the number of layers increases, however, the efficiency and accuracy of the classical BEM decrease significantly. This multi-region method also has the disadvantage in effective and accurate calculations of the elastic fields across internal material interfaces. Thirdly, the fundamental singular solutions, used as the kernel functions, in the classical BEMs vary rapidly if an internal point is very close to the elements located at material interfaces.

### 1.4. Aim and approach of this study

In this paper, a single-region BEM is developed and presented for the analysis of the elastic responses of a non-horizontally layered halfspace subject to tractions (Fig. 1). The generalized Kelvin solution in a multilayered elastic solid given by Yue [12] is used to eliminate the discretization task at the internal interfaces of layered materials. Furthermore, the infinite boundary element technique proposed by Moser et al. [11] is used to take into account the influence of a far-field region because of its straightforward implementation. Other infinite boundary element techniques can be found in the publications by Waston [13], Beer et al. [14], Beer and Waston [15], Zhang et al. [16], Liu and Farris [17], Pereira and Parreira [10], Davies et al. [18], Bu [19], Gao and Davies [20], Moser et al. [11], Salvadori [21], Liang and Liew [22] and Ribeiro et al. [23]. Thirdly, the discontinuous boundary element technique is adopted to deal with the step-discontinuity of material properties at the interface strike line on the horizontal boundary surface of the non-horizontally layered halfspace. Fourthly, special attentions are given to various singular integrals involved in the discretized boundary integral equations. The proposed BEM is applied to specifically solving the elastic response of a halfspace with two or three non-horizontal layers under a square footing loading on the horizontal boundary surface. Numerical results show the influence of non-horizontally
layered materials to the elastic displacement and stress fields induced by the normal footing tractions on the horizontal boundary surface.

## 2. The governing equations for BEM in non-horizontally layered halfspace

Fig. 2 shows the horizontally oriented boundary surface of the non-horizontally layered halfspace model. The interface strike line represents an intersection line of an internal interface plane of any two fully contacted dissimilar material layers with the horizontal boundary surface. The boundary surface is divided into two parts $S_{F}$ and $S_{I}$. They represent a core region around the traction area and a far-field region beyond the traction area, respectively. Accordingly, using the generalized Kelvin solution of a multilayered solid occupying the full-space [12], the boundary integral equations for the non-horizontally layered halfspace without body forces can be expressed as
$c_{i j}(P) u_{j}(P)+\int_{S_{F}+S_{I}} t_{i j}^{Y}(P, Q) u_{j}(Q) d S(Q)=\int_{S_{F}+S_{I}} u_{i j}^{Y}(P, Q) t_{j}(Q) d S(Q)$
where $P$ and $Q$ are, respectively, the field and source points; $t_{j}$ and $u_{j}$ are, respectively, tractions and displacement; $t_{i j}^{Y}$ and $u_{i j}^{Y}$ are, respectively, the kernel functions of the tractions and displacements of the generalized Kelvin solution.

The free term $c_{i j}(P)$ in Eq. (1) depends only upon the asymmetric behavior of the singular terms of the generalized Kelvin solution and the local geometry of the boundary at the point $P$. In using the fundamental solution of a layered space, $c_{i j}(P)=0.5 \delta_{i j}$ for the point $P$ located on a smooth boundary and not at the material interface [24]. When the point $P$ is located at the strike line of the material interface on the horizontal boundary surface, there is no closed-form expression for $c_{i j}(P)$ available in the open literature. So, instead, the discontinuous boundary element technique is adopted for resolving this task. Its details are given in Section 4.1.

After obtaining the displacements and tractions on the boundary, the displacements at any internal point $p$ can be determined by using the displacement integral equations as follows

$$
\begin{equation*}
u_{i}(p)+\int_{S_{F}+S_{I}} t_{i j}^{Y}(p, Q) u_{j}(Q) d S(Q)=\int_{S_{F}+S_{I}} u_{i j}^{Y}(p, Q) t_{j}(Q) d S(Q) \tag{2}
\end{equation*}
$$

By using Eq. (2), the strain-displacement equations and the constitutive equations, the stresses at any internal point $p$ can be expressed as

$$
\begin{equation*}
\sigma_{i j}(p)=\int_{S_{F}+S_{I}} U_{i j k}^{Y}(p, Q) t_{k}(Q) d S(Q)-\int_{S_{F}+S_{I}} T_{i j k}^{Y}(p, Q) u_{k}(Q) d S(Q) \tag{3}
\end{equation*}
$$

where $U_{i j k}^{Y}$ and $T_{i j k}^{Y}$ are the new kernel functions obtained from the displacements and stresses of the generalized Kelvin solution in a layered solid of full-space extent and the relative functions are presented in Appendix A.

## 3. The generalized Kelvin solution in non-horizontally layered fullspace

### 3.1. Two Cartesian coordinate systems and relationship

As shown in Figs. 1 and 2, the boundary integral equations are established in the Cartesian coordinate system Oxyz, where the non-horizontally layered material interface planes have a dip angle $\theta$ to the horizontal xy plane. On the other hand, another Cartesian coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$, shown in Fig. 3, is used for calculating the displacements and stresses of the generalized Kelvin solution, where the $O^{\prime} x^{\prime} y^{\prime}$ plane is parallel and the $z$ axis is perpendicular to the non-horizontally layered material interface planes. The $y$ and $y^{\prime}$ axes of the two Cartesian coordinate systems are parallel to each other.
3.2. The generalized Kelvin solution in $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ and Oxyz coordinates

As shown in Fig. 3, the generalized Kelvin solution (i.e., the kernel functions in Eqs. (1)-(3)) is presented in the Cartesian coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$. The solution was given by Yue [12] for the elastic response of a $n$-layered solid of full-space extent subject to a point body force vector $\left(f_{x^{\prime}}, f_{y^{\prime}}, f_{z^{\prime}}\right)$ at any point in the interior of the layered solid. More details of the solution and its mathematical formulation and verification can be found in Yue [8,9]. The solution given by Yue [8,9,25] is re-named as Yue's solution by Merkel et al. [26] and Maloney et al. [27] in their analysis of the stresses and deformation induced in cells.

Since 2000, Yue and his co-workers [28] have incorporated this generalized Kelvin solution into the classical BEMs for the analysis of the fracture mechanics in layered solids and found
the solutions for many specific problems of interests in science and technology, where the layered material interface planes are horizontally placed and parallel to the horizontal boundary surface. In this paper, we attempt to extend the BEMs for the elastic analysis of displacement and stress fields in the non-horizontally layered halfspace due to tractions on the horizontal boundary surface.

For an ease of reference, some basic formulations of the fundamental solution are provided in Appendix B. It is the analytical solution of closed-form singularity terms for the elastostatic field in a layered solid of full-space extent due to the action of a point force vector (Fig. 3). The total number of the dissimilar layers is an arbitrary nonnegative integer $n$. The dissimilar homogeneous layers adhere an elastic solid of upper halfspace extent and another elastic solid of lower halfspace extent. The interface between any two connected dissimilar layers is fully bonded.

As shown in Fig. 4, the generalized Kelvin solution (i.e., the displacements $u_{i^{\prime}}$ and stresses $\sigma_{i j^{\prime}}$ ) due to the point force vector $\left(f_{x}, f_{y}, f_{z}\right)$ in the Cartesian coordinate system Oxyz can be obtained from those due to the point force vector $\left(f_{x^{\prime}}, f_{y^{\prime}}, f_{z^{\prime}}\right)$ in the Cartesian coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$. Detailed descriptions are presented in Appendix C.

## 4. Five boundary element techniques for BEM in non-horizontally layered halfspace

### 4.1. General

As shown in Fig. 2, the horizontal boundary surface ( $-\infty<x, y<+\infty, z=0$ ) can be divided into two regions $S_{F}$ and $S_{I}$. Each region can be further divided into two sub-areas. The first sub-area has homogeneous material properties and is away from the material interface strike line. The second sub-area has step-discontinuous in material properties along the $x$-axis across the strike line and covers or adjacent to the strike line. So, the entire horizontal boundary surface has to be divided into four types of boundary areas: finite homogeneous area, finite area with strike line, infinite homogeneous area, and infinite area with strike line.

The free term $c_{i j}(P)$ depends on the shape of the boundary and/or elastic parameters. The closed-form expression of $c_{i j}(P)$ is not available for the point $P$ located on an interface strike line where the two elastic parameters are step-discontinuous along the $x$-axis direction on the boundary surface. Secondly, if the nodes of a continuous boundary element are not located on the interface strike line, the line can be covered by the element. For this situation, the Kutt's numerical quadrature presented in Section 5 below cannot be used to calculate the strongly singular integral of the BEM. Thirdly, when the tractions of a node on the interface strike line are unknown, a step-discontinuity in the traction can occur at that node, which implies that the number of equations at that node is smaller than the number of unknowns at that node. Hence, the following discontinuous boundary element technique is adopted to numerically resolve this task for BEM in non-horizontally layered halfspace. Accordingly, the following five boundary element techniques are used.

### 4.2. Continuous finite element for homogenous boundary area (type I)

The conventional eight-node isoparametric and continuous elements which are introduced by Xiao and Yue [28] are shown in Fig. 5a. They are employed to discretize the finite core region $S_{F}$ that is homogeneous in material properties and away from the material interface strike lines. The coordinates at any point in a finite boundary element can be related to its element nodal values as follows:
$x=N x^{e}$
where $x\left(=\left[\begin{array}{lll}x & y & z\end{array}\right]^{\mathrm{T}}\right)$ are Cartesian coordinates of the point with intrinsic coordinates $(\xi, \eta), \boldsymbol{x}^{e}\left(=\left[\begin{array}{llll}\boldsymbol{x}_{1}^{\mathrm{T}} & \boldsymbol{x}_{2}^{\mathrm{T}} & \cdots & \boldsymbol{x}_{8}^{\mathrm{T}}\end{array}\right]^{\mathrm{T}}\right)$ is a vector containing coordinates of nodal points at element $e$ and $N\left(=\left[\begin{array}{llll}N_{1} \mathbf{I}_{3} & N_{2} \mathbf{I}_{3} & \cdots & N_{8} \mathbf{I}_{3}\end{array}\right]\right)$ is a matrix of the shape functions of a continuous finite element listed in Appendix D. 1 ( $\mathbf{I}_{3}$ is a unit matrix ( $3 \times 3$ )).

In a finite boundary element, the displacements and tractions are approximated using the following representation:
$u=N u^{e}$
$t=N t^{e}$
where $\boldsymbol{u}\left(=\left[\begin{array}{lll}u_{x} & u_{y} & u_{z}\end{array}\right]^{\mathrm{T}}\right)$ and $\boldsymbol{t}\left(=\left[\begin{array}{lll}t_{x} & t_{y} & t_{z}\end{array}\right]^{\mathrm{T}}\right)$ are, respectively, the displacements and tractions of the point with intrinsic coordinates $(\xi, \eta)$ and $\boldsymbol{u}^{e}\left(=\left[\begin{array}{llll}\boldsymbol{u}_{1} & \boldsymbol{u}_{2}{ }^{\mathrm{T}} & \cdots & \boldsymbol{u}_{8}{ }^{\mathrm{T}}\end{array}\right]^{\mathrm{T}}\right)$ and $\boldsymbol{t}^{\boldsymbol{e}}\left(=\left[\begin{array}{llll}\boldsymbol{t}_{1}^{\mathrm{T}} & \boldsymbol{t}_{2}{ }^{\mathrm{T}} & \cdots & \boldsymbol{t}_{8}{ }^{\mathrm{T}}\end{array}\right]^{\mathrm{T}}\right)$ are, respectively, vectors containing displacements and tractions of nodal points at element $e$.

### 4.3. Discontinuous finite element for boundary area with interface strike line (type II)

The eight-node isoparametric and discontinuous elements (type II), shown in Fig. 5b, are employed to discretize the finite area with strike line on the region $S_{F}$. The element type II is positioned on two sides of the material interface strike line. Comparing to the element type I, the nodes 1,2 and 5 of the discontinuous element are moved inside to avoid the discontinuities of unknown tractions and the traction kernel. The shape functions of the element type II are listed in Appendix D. 2 where the discontinuous nodes are at a distance of $1 / 3$ from the element edge $\eta=-1$. The coordinates, displacements and stresses at any point in a finite boundary element can be related to its element nodal values by using Eqs. (11a)-(11c).

### 4.4. Continuous infinite element for homogenous boundary area (type III)

In order to consider the influence of a far field, the continuous infinite element developed by Moser et al. [11], as shown in Fig. 6a, is employed to the infinite region $S_{I}$ that is homogeneous in material properties and away from the material interface strike line. The coordinates at any point in an infinite element is then related to its element nodal coordinates as follows:
$\boldsymbol{x}={ }^{\infty} \boldsymbol{N} \boldsymbol{x}^{e}$
where ${ }^{\infty} \boldsymbol{N}\left(=\left[\begin{array}{llll}{ }^{\infty} N_{1} \mathbf{I}_{3} & { }^{\infty} N_{2} \mathbf{I}_{3} & \cdots & { }^{\infty} N_{6} \mathbf{I}_{3}\end{array}\right]\right)$ is a matrix of the shape functions of an infinite element listed in Appendix D. 3 ( $\mathbf{I}_{3}$ is a unit matrix ( $3 \times 3$ )).

In an infinite boundary element, the displacements and tractions are approximated using the following representation:
$\boldsymbol{u}={ }^{\infty} \boldsymbol{N}^{u} \boldsymbol{u}^{e}$
$\boldsymbol{t}={ }^{\infty} \boldsymbol{N}^{t} \boldsymbol{t}^{e}$
where ${ }^{\infty} \boldsymbol{N}^{u}\left(=\left[\begin{array}{llll}{ }^{\infty} N_{1}^{u} \mathbf{I}_{3} & { }^{\infty} N_{2}^{u} \mathbf{I}_{3} & \ldots & { }^{\infty} N_{6}^{u} \mathbf{I}_{3}\end{array}\right]\right)$ and ${ }^{\infty} \boldsymbol{N}^{t}\left(=\left[\begin{array}{llll}{ }^{\infty} N_{1}^{t} \mathbf{I}_{3} & { }^{\infty} N_{2}^{t} \mathbf{I}_{3} & { }^{\circ} & { }^{\infty} N_{6}^{t} \mathbf{I}_{3}\end{array}\right]\right)$ are, respectively, matrices of the interpolation functions of displacements and tractions listed in Appendix D. 3 ( $\mathbf{I}_{3}$ is a unit matrix $(3 \times 3)$ ).

### 4.5. Discontinuous infinite element for boundary area with interface strike line (type IV)

The discontinuous infinite elements (type IV), shown in Fig. 6b, are employed to discretize the infinite area with strike line on the region $S_{I}$. Comparing to the element type III, the nodes 1 and 4 of the discontinuous element are moved inside to avoid the discontinuity of the traction kernel. The shape functions and interpolation functions of the element type IV are listed in Appendix D. 4 where the discontinuous nodes are at a distance of $1 / 3$ from the element edge $\xi=-1$. The coordinates, displacements and stresses at any point in an infinite boundary element can be related to its element nodal values by using Eqs. (12a)-(12c).

### 4.6. Discontinuous infinite element for boundary area with interface strike line (type V)

The discontinuous infinite elements (type V), shown in Fig. 6c, are also employed to discretize the infinite area with strike line on the region $S_{I}$. Comparing to the element type III, the nodes 2 and 3 of the discontinuous element are moved inside to avoid the discontinuity of the traction kernel. The shape functions and interpolation functions of the element type V are listed in Appendix D. 5 where the discontinuous nodes are at a distance of $1 / 3$ from the element edge $\xi=1$. The shape functions and interpolation functions are different from those of elements (type IV) and substituting these functions of the element type V into Eqs. (12a)-(12c), the coordinates, displacements and tractions at any point in the element type V can be related to its element nodal values.

## 5. Numerical integrations for BEM in non-horizontally layered halfspace

### 5.1. Discretized governing boundary integral equations

The above five boundary elements are used for discretization of the horizontal boundary surface $S_{F}$ and $S_{I}$. The governing boundary integral equation (1) can be re-written as the following discretized expressions

$$
\begin{array}{r}
c_{i j}(P) u_{j}(P)+\sum_{e=1}^{F B E} \int_{S_{F}} t_{i j}^{Y}(P, Q) u_{j}(Q) d S(Q)+\sum_{e=1}^{I B E} \int_{S_{i}^{l}} t_{i j}^{Y}(P, Q) u_{j}(Q) d S(Q)  \tag{13}\\
=\sum_{e=1}^{F B E} \int_{S_{F}^{s}} u_{i j}^{Y}(P, Q) t_{j}(Q) d S(Q)+\sum_{e=1}^{I B E} \int_{S_{i}^{L}} u_{i j}^{Y}(P, Q) t_{j}(Q) d S(Q)
\end{array}
$$

where $F B E$ and $I B E$ are the numbers of finite and infinite boundary elements, respectively.
Using Eqs. (11) and (12), Eq. (13) can be further re-written as follows:

$$
\begin{array}{r}
c_{i j}(P) u_{j}(P)+\sum_{e=1}^{F B E} \sum_{\alpha=1}^{8} u_{j}^{\alpha}\left(Q^{\alpha}\right) \int_{S_{F}^{S}} t_{i j}^{Y}(P, Q) N_{\alpha} d S(Q)+\sum_{e=1}^{I B E} \sum_{\alpha=1}^{6} u_{j}^{\alpha}\left(Q^{\alpha}\right) \int_{S_{i}^{t}} t_{i j}^{Y}(P, Q)^{\infty} N_{\alpha}^{u} d S(Q) \\
=\sum_{e=1}^{F B E} \sum_{\alpha=1}^{8} t_{j}^{\alpha}\left(Q^{\alpha}\right) \int_{S_{F}^{e}} u_{i j}^{Y}(P, Q) N_{\alpha} d S(Q)+\sum_{e=1}^{I B E} \sum_{\alpha=1}^{6} t_{j}^{\alpha}\left(Q^{\alpha}\right) \int_{S_{T}^{S}} u_{i j}^{Y}(P, Q)^{\infty} N_{\alpha}^{t} d S(Q) \tag{14}
\end{array}
$$

The coefficients of the displacements and tractions at all the nodes in Eq. (14) are first calculated with numerical integration techniques. A set of linear algebraic equations is then formed for the determination of all boundary unknowns of displacements and tractions at the nodes. The generalized minimum residual algorithm is adopted in solving the set of linear algebraic equations, which can reduce the calculation time.

### 5.2. Numerical techniques for regular and weakly singular integrals

In Eq. (14), when the source point $P$ is not collocated on an integration element within either $S_{F}^{e}$ or $S_{I}^{e}$, all the integrals are regular. These regular integrals can be evaluated using the Gaussian quadrature rule. When the source point $P$ is located in the same boundary element as the field point $Q$ and $P=Q^{\alpha}$, both weakly and strongly singular integrals exist. An element subdivision technique developed by Lachat and Watson [29] is used to calculate the weakly singular integrals. The strongly singular integrals are evaluated using the numerical method described in Section 5.3.

After the unknown displacements and tractions at the nodes are determined from Eq. (14), the discretized equations (2) and (3) do not contain unknown variables. Because the internal point $p$ under consideration is located within the domain, the singular integrals in Eqs. (2) and
(3) do not exist. Thus, the traditional Gaussian quadrature can be directly used for calculating the displacements and stresses at any internal point. When the internal point $p$ is very close to the integration region on the boundary surface, the kernels in Eqs. (2) and (3) can vary rapidly. To obtain the desired accuracy of such integrals, an effective method is to implement a scheme of subdividing the element into sub-regions of integration. The subdivision scheme in Beer et al. [30] and Gao and Davies [31] are adopted in the present BEM.

### 5.3. Numerical techniques for strongly singular integrals

### 5.3.1 General

Since infinite boundary elements are used in this BEM, the strongly singular integrals cannot be evaluated indirectly using the rigid displacement method. Beer and Watson [14] evaluated the strongly singular integrals indirectly through the computation of an azimuthal integral, which is not an easy task. Pereira and Parreira [10] and Moser et al. [11] used the method proposed by Guiggiani and Gigante [32] to directly evaluate the strongly singular integrals. More recently, Gao [33] and Gao et al. [34] analytically evaluated all kinds of singular boundary integrals in 2D/3D BEM analysis. However, this direct method may not be applicable to the BEM of fundamental solutions with complex functions. If a numerical quadrature can be used directly for the evaluation of strongly singular integrals, the implementation of infinite boundary elements becomes a straightforward task.

### 5.3.2 Kutt's numerical quadrature

A Kutt's numerical quadrature is adopted to calculate the strongly singular integrals for the infinite and finite elements, shown in Figs. 5 and 6. Here, the proposed numerical quadrature is presented only for the continuous infinite element (type III), shown in Fig. 6a and, of course, is easily used to other types of boundary elements, as shown in Figs. 5 and 6. On an infinite boundary element, the strongly singular integral can be expressed as

$$
\begin{equation*}
\int_{-1}^{1} \int_{-1}^{1} t_{i j}^{Y}\left[P\left(\xi^{c}, \eta^{c}\right), Q(\xi, \eta)\right]^{\infty} N_{\alpha}^{u}(\xi, \eta) J(\xi, \eta) d \xi d \eta \tag{15}
\end{equation*}
$$

where $J$ is the Jacobian transformation. The collocation point $P\left(\xi^{c}, \eta^{c}\right)$ in Eq. (15) coincides with one of the nodal points on the element.

Following Pan and Yuan [35], the strongly singular integral (15) can also be calculated by using the numerical quadrature proposed by Kutt $[36,37]$. Introducing the following polar point at the source point $P$ on the element
$\xi=\xi^{c}+r \cos \theta$ and $\eta=\eta^{c}+r \sin \theta$
Eq. (15) can then be written as

$$
\begin{equation*}
\sum_{m} \int_{\theta_{1}}^{\theta_{2}} \int_{0}^{R(\theta)} t_{i j}^{Y}\left[P\left(\xi^{c}, \eta^{c}\right), Q(r, \theta)\right]^{\infty} N_{\alpha}^{u}(r, \theta) J(r, \theta) r d r d \theta \tag{17}
\end{equation*}
$$

where the summation on $m$ is for all the triangles on the element.
These triangles are formed by connecting the source point and two corners. In the ensuing, we present this subdivision technique for a continuous infinite boundary element, as shown in Fig. 7. The details are further given below on the basis of three locations of the collocation point.

## Location a) (Fig. 7a)

If the collocation point is located at the corner node 1 of the element, the element is then divided into two triangles and the summation on $m$ in Eq. (17) is from 1 to 2. As shown in Fig. 7a, $\left(\xi^{c}, \eta^{c}\right)=(-1,-1)$. Eq. (16) can be re-written as $\xi=-1+r \cos \theta$ and $\eta=-1+r \sin \theta$. For the triangle 1, $\theta \in[0, \pi / 4]$ and $R(\theta)=2 / \cos \theta$; for the triangle $2, \theta \in[\pi / 4, \pi / 2]$ and $R(\theta)=2 / \cos (\theta-\pi / 2)$.

## Location b) (Fig. 7b)

If the collocation point is located at the corner node 2, the element needs to be divided into two triangles, and the summation on $m$ in Eq. (17) is from 1 to 2. As shown in Fig. 7b, $\left(\xi^{c}, \eta^{c}\right)=(1,-1)$. Eq. (16) can be re-written as $\xi=1+r \cos \theta$ and $\eta=-1+r \sin \theta$. For the triangle $1, \quad \theta \in[\pi / 2,3 \pi / 4]$ and $R(\theta)=2 / \cos (\theta-\pi / 2)$; for the triangle 2 , $\theta \in[3 \pi / 4, \pi]$ and $R(\theta)=2 / \cos (\pi-\theta)$.

## Location c) (Fig. 7c)

If the collocation point is located at the middle node 5 of the square side, the element needs to be divided into three triangles, and the summation on $m$ in Eq. (17) is from 1 to 3. As shown in in Fig. $7 \mathrm{c},\left(\xi^{c}, \eta^{c}\right)=(0,-1)$. Eq. (16) can be re-written as $\xi=r \cos \theta$ and $\eta=-1+r \sin \theta$. For different triangles, there are the following integral parameters:

- For the triangle $1, \theta \in[0, \operatorname{atan}(2)]$ and $R(\theta)=1 / \cos \theta$;
- For the triangle 2, $\theta \in[\operatorname{atan}(2), \pi-\operatorname{atan}(2)]$ and $R(\theta)=2 / \cos (\theta-\pi / 2)$;
- For the triangle 3, $\theta \in[\pi / 2-\operatorname{atan}(2), \pi]$ and $R(\theta)=1 / \cos (\pi-\theta)$.

From Eq. (17), it is observed that the integrand is $O(1 / r)$, where $r$ is the distance between the source and field points. Therefore, the Kutt's numerical quadrature can be utilized to evaluate the inner finite-part integral-part integral with respect to $r$. The outer integral with respect to $\theta$ is regular and can be calculated with the regular Gauss quadrature. For a given Gauss point $\theta_{j}$, the inner integral in Eq. (17) can be approximated with Kutt's $N$-point equispace quadrature as follows

$$
\begin{equation*}
\int_{0}^{R} \frac{f(r)}{r} d r \approx \sum_{l=1}^{N}\left(w_{l}+c_{l} \ln R\right) f\left(\frac{l-1}{N} R\right) \tag{18}
\end{equation*}
$$

where $w_{l}$ are the weights and $c_{l}$ is the coefficients given by Kutt, and the integrand is given by

$$
\begin{equation*}
f(r)=t_{i j}^{Y}\left[P\left(\xi^{c}, \eta^{c}\right), Q\left(r, \theta_{j}\right)\right]^{\infty} N_{\alpha}^{u}\left(r, \theta_{j}\right) J\left(r, \theta_{j}\right) r^{2} \tag{19}
\end{equation*}
$$

In applying Kutt's $N$-point equispace quadrature (18), it is assumed that the integrand $f(r) \in C^{0}[0, R]$ and $f(r) \in C^{1}$ in the neighborhood of $r=0$. In discretizing the boundary surface, discontinuous finite and infinite elements are positioned on the two sides of the material interface strike line. In this case, the integrand $f(r)$ in Eq. (19) is smoothly continuous for the domain $[0, R]$ and the quadrature (18) is used for evaluating the integral (17). In the following numerical examples, the Kutt's 20-point equispace quadrature is used in
the finite-part integral with respect to $r$, and 20 Gaussian points for the regular outer integral with respect to $\theta_{j}$.

## 6. Numerical verifications

Based on the above analytical and numerical equations, the BEM program in Fortran is made to calculate the elastostatic fields of displacements and stresses in a non-horizontally layered halfspace. This BEM program can calculate all the field variables at any point in a non-homogeneous layered halfspace induced by different tractions on the horizontal boundary surface. This section presents the numerical results of two elastic halfspace models to verify the present BEM program.

### 6.1. Square footing on a homogeneous halfspace (example 1)

This example in Moser et al. [11] is used to verify the present BEM program. The elastic halfspace has the modulus of $10000 \mathrm{kN} / \mathrm{m}^{2}$ and a zero Possion's ratio. A pressure of 100 $\mathrm{kN} / \mathrm{m}^{2}$ uniformly acts on a square area with the side length of 2 m on the horizontal boundary surface of the elastic halfspace. The boundary element mesh given in Fig. 4 of Moser et al. [11] is used. The present BEM program gives the vertical displacements of 0.022785 and 0.011379 m at the center and the corner of the loaded square area, respectively. The exact vertical displacements at these two points are 0.022444 and 0.011222 m , respectively. The maximum relative difference between the present numerical results and the exact results is 1.52\%.

### 6.2. Square footing on a bi-material halfspace with non-horizontal interface (example 2)

The example in Pereira and Parreira [10] is also used to verify the present BEM program. As shown in Fig. 8, an elastic halfspace has two different materials that are fully bonded at their non-horizontal interface. The dip angle of the non-horizontal interface plane is $45^{\circ}(\pi / 4)$. The strike line of the inclined material interface plane with the horizontal boundary surface is located at the line of $x=1.5 \mathrm{~m},-\infty<y<+\infty$ and $z=0$. A square area ( $1 \mathrm{~m} \times 1 \mathrm{~m}$ ) of its horizontal boundary surface is subject to a uniformly distributed footing pressure of 100 $\mathrm{kN} / \mathrm{m}^{2}$. The elastic moduli of the two materials are, respectively, 10000 and $2000 \mathrm{kN} / \mathrm{m}^{2}$. The
values of both materials' Poisson's ratios are zero.
As shown in Fig. 9, the core region $S_{F}$ is represented by a square area ( $5 \mathrm{~m} \times 5 \mathrm{~m}$ ) covering the loaded square area ( $1 \mathrm{~m} \times 1 \mathrm{~m}$ ). The center of the square core region $S_{F}$ is identical to the center of the loaded square area. The square core region is further divided into the first and second sub-areas. The first sub-area is not adjacent to the interface strike line and is discretized with a total of 284 finite continuous elements (type I). The second sub-area is on both sides of the interface strike line and is discretized with a total of $36(2 \times 18)$ finite discontinuous elements (type II). The far-field region $S_{I}$ outside the square core region is also further divided into the third and fourth sub-areas. The third sub-area is not adjacent to the interface strike line and is discretized with a total of 60 infinite continuous elements (type III). The fourth sub-area is on both sides of the interface strike line. Both side areas are discretized with one infinite discontinuous element (type IV) and one infinite discontinuous element (type V). The discontinuous elements are positioned on the narrow left and right side areas adjacent the interface strike line.

The present BEM program gives the vertical displacement values of $0.022900,0.011405$ and 0.011233 m at the center, the two corners nearer to the interface and the other two corners of loaded square area, respectively. Because of symmetry, the vertical displacements at the two corners ( $x=1 \mathrm{~m}, y= \pm 1 \mathrm{~m}, \mathrm{z}=0$ ) are the same. The corresponding values presented by Pereira and Parreira [10] are $0.022428,0.011216$ and 0.011195 m , respectively. The maximum relative difference between these two BEM programs is $2.1 \%$. Pereira and Parreira [10] used the classical Kelvin's solution based BEM program where the inclined interface plane is discretized with elements. The present BEM does not need to discretize the inclined interface plane.

## 7. Numerical examples

### 7.1. Square footing on bi-material halfspace with one non-horizontal interface

### 7.1.1 General

This elastostatic problem is shown in Fig. 10. It is similar to the elastostatic problem examined in Section 6.2 and by Pereira and Parreira [10] (Fig. 8). The non-horizontal
bi-material interface dips at the arbitrary angle $\theta$ with the horizontal boundary surface. Its strike line is exactly along the $y$-axis. The bi-materials have their elastic properties as follows: $E_{1}=20 \mathrm{GPa}, v_{1}=0.3, E_{2}=40 \mathrm{GPa}$, and $v_{2}=0.25$. The square loading area has a side length of 2 m and is acted by a uniformly distributed pressure of 100 MPa . The adopted boundary element meshes are much similar to the meshes shown in Fig. 9. The discontinuous elements are positioned on the two narrow zones on both sides of the material interface strike line (i.e., the $y$-axis). The two locations (noted as $a$ and $b$ in Fig. 10) correspond to ( $x=-1 \mathrm{~m}, z=0$ ) and ( $x=1 \mathrm{~m}, z=0$ ), respectively. The symbol $c^{\prime}$ is the horizontal distance from the point $c$ on the material interface to the vertical coordinate plane ( $x=0,-\infty<y, z<+\infty$ ). The $c^{\prime}$ values are equal to $1.7321,0.5774,0,-0.5774$ and -1.7321 m , respectively for $\theta=30^{\circ}, 60^{\circ}$, $90^{\circ}, 120^{\circ}$ and $150^{\circ}$. For benchmarking of the numerical results, the elastostatic problems of the homogeneous halfspace with either the material $1\left(E_{1}=20 \mathrm{GPa}\right.$ and $\left.v_{1}=0.3\right)$ or the material $2\left(E_{2}=40 \mathrm{GPa}\right.$ and $\left.v_{2}=0.25\right)$ under the same pressure loading are also calculated and presented below.

### 7.1.2 Variations of elastic fields with $x$-axis for different $\theta$ values

Fig. 11 shows the variations of the three displacements $\left(u_{x}, u_{y}, u_{z}\right)$ along $-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m}$ at ( $y=0.5 \mathrm{~m}$ and $z=1 \mathrm{~m}$ ) for the dip angle $\theta$ value of $30^{\circ}, 60^{\circ}, 90^{\circ}, 120^{\circ}$, or $150^{\circ}$. It is evident that (1) the $u_{x}$ in a homogeneous elastic halfspace shows an anti-symmetrical distribution about the vertical coordinate plane $(x=0,-\infty<y, z<+\infty)$ and (2) the $u_{y}$ and $u_{z}$ in a homogeneous elastic halfspace shows a symmetrical distribution about the vertical coordinate plane. However, the ( $u_{x}, u_{y}, u_{z}$ ) in the bi-material elastic halfspace do not have such anti-symmetrical or symmetrical distributions. The $u_{z}$ (or $u_{y}$ ) value in the bi-material halfspace for each $\theta$ value is almost bonded by the two corresponding $u_{z}$ (or $u_{y}$ ) values in the two homogeneous elastic halfspaces. But, the $u_{x}$ value in the bi-material halfspace can
be either larger or less than the two corresponding $u_{x}$ values in the two homogeneous elastic halfspaces. The $u_{x}$ in the bi-material halfspace increases as $\theta$ increases from $30^{\circ}$ to $90^{\circ}$ and decreases as $\theta$ increases from $90^{\circ}$ to $150^{\circ}$. The $u_{y}$ and $u_{z}$ in the bi-material halfspace increase as $\theta$ increases from $30^{\circ}$ to $150^{\circ}$. Such results are due to the fact that as $\theta$ increases, the softer material $1\left(E_{1}=20 \mathrm{GPa}\right.$ and $\left.v_{1}=0.3\right)$ is enlarged whilst the harder material $2\left(E_{2}=40\right.$ GPa and $v_{2}=0.25$ ) is reduced. Furthermore, except the $u_{y}$ at $\theta=60^{\circ}$ in Fig. 11b, all the $\left(u_{x}, u_{y}, u_{z}\right)$ in Fig. 11 show a very-slightly non-smooth continuation across the interface plane.

Fig. 12 shows the variations of the six stresses ( $\sigma_{x x}, \sigma_{x y}, \sigma_{x z}, \sigma_{y y}, \sigma_{y z}, \sigma_{z z}$ ) along $-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m}$ at $(y=0.5 \mathrm{~m}$ and $z=1 \mathrm{~m})$ for the dip angle $\theta$ value of $30^{\circ}, 60^{\circ}, 90^{\circ}, 120^{\circ}$, or $150^{\circ}$. Table 1 presents the stress jumps across the material interface plane. The effects of the bi-materials to the six stress values are evident comparing to these of the two homogeneous elastic halfspaces. In particular, $\sigma_{x x}, \sigma_{x y}$ and $\sigma_{x z}$ are discontinuous across the material interface plane for $\theta=30^{\circ}, 60^{\circ}, 120^{\circ}$ or $150^{\circ} . \sigma_{y y}, \sigma_{y z}$ and $\sigma_{z z}$ are always discontinuous across the material interface plane for $\theta=30^{\circ}, 60^{\circ}, 90^{\circ}, 120^{\circ}$ or $150^{\circ}$. The corresponding stress jumps are between 0.1489 MPa and 41.1799 MPa . At some interface points, some stresses (such as $\sigma_{x x}, \sigma_{x y}, \sigma_{x z}, \sigma_{y z}, \sigma_{z z}$ ) can have some very high values comparing to the values at other locations and the results of the homogeneous halfspace models.

Furthermore, the BEM results for the jumps ( $\Delta \sigma_{i j}=\left|\sigma_{i j}\left(x=c^{\prime+}\right)-\sigma_{i j}\left(x=c^{\prime-}\right)\right|$ ) of $\sigma_{x x}, \sigma_{x y}$ and $\sigma_{x z}$ values at the interface plane for $\theta=90^{\circ}$ are between 0.0001 MPa and 0.0009 MPa (Table 1) and very close to the theoretical zero (0) value since the three stresses are non-smoothly continuous at the interface plane [8, 9, 25]. Such near zero stresses jumps for $\theta=90^{\circ}$ can indicate that the proposed BEM method is of high accuracy.
7.1.3 Elastic fields on the horizontal plane $\mathrm{z}=1 \mathrm{~m}$ for $\theta=60^{\circ}$

Fig. 13 shows the variations of $\left(u_{x}, u_{y}, u_{z}\right)$ on the horizontal plane area ( $-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m}, 0 \leq y \leq 5 \mathrm{~m}$ ) at $z=1 \mathrm{~m}$ for $\theta=60^{\circ}$, respectively. As shown in Fig. 11a, $u_{x}$ has a slightly non-smooth continuation at the point ( $x=c^{\prime}=0.5774 \mathrm{~m}, y=0.5 \mathrm{~m}, \mathrm{z}=1 \mathrm{~m}$ ) on the material interface. Fig. 13a further shows that $u_{x}$ has a slightly non-smooth continuation across the material interface line of ( $x=c^{\prime}=0.5774 \mathrm{~m}, 0 \leq y \leq 5 \mathrm{~m}, z=1 \mathrm{~m}$ ). Fig. 13a and 13c show the both $u_{x}$ and $u_{z}$ decrease as $y$ increases from 0 to $5 \mathrm{~m} . u_{y}$ is zero along the symmetrical line $y=0$ and $-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m} . u_{y}$ has two peak values and increases at the beginning and then decrease as $y$ increases. The displacement $u_{y}$ is non-smoothly continuous at the point ( $x=c^{\prime}=0.5774 \mathrm{~m}, 0 \leq y \leq 0.5 \mathrm{~m}, z=1 \mathrm{~m}$ ) on the material interface.

Fig. 14 shows the variations of the six stresses ( $\sigma_{x x}, \sigma_{x y}, \sigma_{x z}, \sigma_{y y}, \sigma_{y z}, \sigma_{z z}$ ) over the above horizontal area ( $-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m}, 0 \leq y \leq 5 \mathrm{~m}$ ) covering the square footing area at $z=1$ m for $\theta=60^{\circ}$. $\sigma_{x y}$ and $\sigma_{y z}$ are zero along the anti-symmetrical line $y=0$. In this way, other stress components except $\sigma_{x y}$ and $\sigma_{y z}$ are discontinuous across the material interface line of $\left(x=c^{\prime}=0.5774 \mathrm{~m}, 0 \leq y \leq 5 \mathrm{~m}, z=1 \mathrm{~m}\right)$. As $y$ increases, the absolute values of $\sigma_{x x}, \sigma_{x z}$ and $\sigma_{z z}$ decrease in general and the absolute values of $\sigma_{x y}, \sigma_{y y}$ and $\sigma_{y z}$ first increase and then decreases.

### 7.1.4 Elastic fields at different z for $\mathrm{y}=0.5 \mathrm{~m}$ and $\theta=60^{\circ}$

Fig. 15 shows the variations of $\left(u_{x}, u_{y}, u_{z}\right)$ with $x(-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m})$ with different $z$ values at $y=0.5 \mathrm{~m}$ and $\theta=60^{\circ}$. As $z$ increases, $\left(u_{x}, u_{y}, u_{z}\right)$ decrease in general. The effect of the inclined bi-material properties is clearly shown in Fig. 15. At the bi-material interface points, $u_{x}$ and $u_{y}$ have clear non-smooth continuation properties whilst $u_{z}$ has less evident non-smooth continuation feature.

Considering the importance of the Mises stress in analyzing the yielding state, Fig. 16 presents the variations of the Mises stress with $x(-5 m \leq x \leq 5 m)$ with different $z$ values at $y=0.5 \mathrm{~m}$ and $\theta=60^{\circ}$. The Mises stress is defined as
$\sigma_{e}=\frac{1}{\sqrt{2}}\left[\left(\sigma_{x x}-\sigma_{y y}\right)^{2}+\left(\sigma_{y y}-\sigma_{z z}\right)^{2}+\left(\sigma_{z z}-\sigma_{x x}\right)^{2}+6\left(\sigma_{x y}^{2}+\sigma_{y z}^{2}+\sigma_{z x}^{2}\right)\right]^{1 / 2}$
It can be found that the Mises stress has step discontinuation at the material interface points. As z increases, the absolute values and the discontinuous values of the Mises stress decrease in general. The effect of the bi-materials with inclined interface can be clearly observed in the non-regular variations of the Mises stress with $x$ and $z$.

### 7.2. Square footing on three material halfspace with two non-horizontal interfaces

### 7.2.1 General

The present BEM program is further used to examine the elastic field in a three-material halfspace with two non-horizontal and parallel interfaces subject to the square footing pressure of 100 MPa (Fig. 17). The three materials have the elastic parameters as follows: $\left(E_{1}=20 \mathrm{GPa}, v_{1}=0.3\right), \quad\left(E_{2}=40 \mathrm{GPa}, v_{2}=0.25\right)$ and $\left(E_{3}=20 \mathrm{GPa}, v_{3}=0.3\right)$, respectively. In other words, the materials 1 and 3 have the same elastic properties and the material 2 is their sandwich.

In Fig. 17a, the points $a$ and $b$ are at $x=-1 \mathrm{~m}$ and 1 m on the boundary urface $(z=0)$, respectively. In Fig. 17b, $c_{1}^{\prime}$ and $c_{2}^{\prime}$ are the horizontal distances from the points $c_{1}$ and $c_{2}$ on the material interfaces to the vertical coordinate plane $O y z$ along the $x$-axis direction, respectively. For $\theta=30^{\circ}, 60^{\circ}, 90^{\circ}, 120^{\circ}$ and $150^{\circ}$, the $c_{1}^{\prime}$ values are equal to $0.7321,0.4226$, $-1,-1.5774$ and -2.7321 m , respectively and the $c_{2}^{\prime}$ values are equal to $2.7321,1.5774,1$, 0.4226 and 0.7321 m , respectively. The adopted boundary meshes are similar to those shown in Fig. 9. The discontinuous elements are positioned on the two sides of each of the two material interface strike lines $(x=1 \mathrm{~m}$ and $z=0)$ and $(x=-1 \mathrm{~m}$ and $z=0)$. The adopted boundary meshes have a total of 320 finite elements (248 type I elements and 72 type II
elements) and a total of 64 infinite elements (56 type III elements, 4 type IV elements and 4 type V elements).

### 7.2.2 Numerical results

Fig. 18 shows the variations of three displacements $\left(u_{x}, u_{y}, u_{z}\right)$ with $x(-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m})$ at $y=0.5 \mathrm{~m}$ and $\mathrm{z}=1 \mathrm{~m}$ for $\theta=30^{\circ}, 60^{\circ}, 90^{\circ}, 120^{\circ}$, or $150^{\circ}$. The results of two homogeneous elastic halfspaces (i.e., $\theta=0^{\circ}$ or $180^{\circ}$ ) are also presented in Fig. 18. The two curves of each component of $\left(u_{x}, u_{y}, u_{z}\right)$ for $\theta=30^{\circ}$ and $\theta=150^{\circ}$ (similarly or $\theta=60^{\circ}$ and $\theta=120^{\circ}$ ) have the symmetrical or anti-symmetrical variations with respect to the $x$ axis, which is consistent to the fact that the materials 1 and 3 have the same elastic properties and each pair of the elastostatic problems are symmetrically identical. The effects of the three materials with inclined two interface planes to the elastic displacements are evidently shown with respect to the results of the two homogeneous halfspaces. The results of the two homogeneous halfspaces can offer the upper or lower bounds for the results of the three-material halfspace models.

Fig. 19 shows the variations of ( $\sigma_{x x}, \sigma_{x y}, \sigma_{x z}, \sigma_{y y}, \sigma_{y z}, \sigma_{z z}$ ) with $x(-5 \mathrm{~m} \leq x \leq 5 \mathrm{~m})$ with different $\theta$ values at $y=0.5 \mathrm{~m}$ and $z=1 \mathrm{~m}$. Table 1 also presents the stress jumps $\left(\Delta \sigma_{i j}=\mid \sigma_{i j}\left(x=c_{1}^{\prime+}\right.\right.$ or $\left.c_{2}^{\prime+}\right)-\sigma_{i j}\left(x=c_{1}^{\prime-}\right.$ or $\left.\left.c_{2}^{\prime-}\right) \mid\right)$ in three-material halfspace for different $z$ across the material interface plane. Similarly the symmetrical and/or anti-symmetrical results are obtained for the six stresses for each pair of the three-material halfspace problem. The effects of the three materials with two inclined interface planes to the elastic stresses are evidently shown with respect to the results of the two homogeneous halfspaces. At each interface point, some stresses (such as $\sigma_{x x}, \sigma_{y y}, \sigma_{y z}$ ) can have some very high values comparing to the values at other locations and the results of the homogeneous halfspace models.

### 7.3 Further analysis of displacements and stresses across material interfaces

The elastic fields in a bi-material or three-material halfspace with non-horizontally oriented interface planes due to tractions are presented and examined in detail. The effects of the
inclined material interface to the elastic displacements and stresses are analyzed with respect to the results of homogeneous elastic halfspace. Across the material interface, the elastic displacements are non-smoothly continuous to different degrees, which is affected by the interface declination angle $\theta$. Across the material interfaces, all the six stress components can have step-discontinuation for $\theta \neq 90^{\circ}$ whereas only the three stresses $\sigma_{y y}, \sigma_{y z}$ and $\sigma_{z z}$ are step-discontinuous for $\theta=90^{\circ}$. Distribution of the stresses in a non-horizontally layered halfspace is different to those in a homogeneous halfspace due to the same loading. In particular, some stress components can have very high values at and adjacent the interface planes, which can be important to tensile or shear failures in non-homogeneous materials.

## 8. Summary and conclusions

This paper adopts the generalized Kelvin solution of a multilayered solid and develops a new BEM program for the effect and efficient analysis and calculation of the elastic field in non-horizontally layered halfspace subject to tractions. The BEM program uses five boundary element techniques to handle the mixtures of the boundary surface conditions. They include the material interface strike line and the infinite extension of both boundary surface areas and the strike lines.

The use of the generalized Kelvin solution can eliminate the requirement of mesh discretization for the internal material interface planes. The only surface that needs to be discretized is the horizontal boundary surface of infinite extent. The infinite boundary elements are used to effectively and efficiently take into account the influence of a far region to the numerical approach. The finite and infinite discontinuous boundary elements are introduced to overcome the numerical problems due to the step-discontinuous material property across the interface strikes on the boundary surface. Consequently, the singular integrals in the governing boundary integral equations are calculated using the existing numerical integration techniques. Numerical results for some typical elastostatic problems presented in the paper indicate that the BEM program can be used for the elastic fields in the non-horizontal layered halfspace.

More detailed investigations need to be carried for accurately enhance our knowledge and
capability for the related problems in science and engineering in the future. The numerical results presented in this paper are due to the completely flexible footing loading. The BEM program can be applied to examine the contact problems where the rigid or elastic footing can be used because of the adoption of the discontinuous elements.

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## Appendix A. The kernel functions $U_{i j k}^{Y}$ and $T_{i j k}^{Y}$

The kernel functions $U_{i j k}^{Y}$ and $T_{i j k}^{Y}$ in Eq. (3) can be described by using the constitutive equations as follows

$$
\begin{align*}
& U_{i j k}^{Y}=\lambda \delta_{i j} U_{m m k}^{\varepsilon}+2 \mu U_{i j k}^{\varepsilon}  \tag{A.1}\\
& T_{i j k}^{Y}=\lambda \delta_{i j} T_{m m k}^{\varepsilon}+2 \mu T_{i j k}^{\varepsilon} \tag{A.2}
\end{align*}
$$

where $\lambda$ and $\mu$ are Lamé constants, and $U_{i j k}^{\varepsilon}$ and $T_{i j k}^{\varepsilon}$ can be written as
$U_{i j k}^{\varepsilon}=\left(u_{i k, j}^{Y}+u_{j k, i}^{Y}\right) / 2$
$T_{i j k}^{\varepsilon}=\left(t_{i k, j}^{Y}+t_{j k, i}^{Y}\right) / 2$
The derivatives of the displacements in Eq. (A.3) are approximated as

$$
\begin{align*}
& \frac{\partial u_{i k}^{Y}}{\partial x} \approx \frac{1}{2 D}\left[u_{i k}^{Y}(x+D, y, z)-u_{i k}^{Y}(x-D, y, z)\right]  \tag{A.5}\\
& \frac{\partial u_{i k}^{Y}}{\partial y} \approx \frac{1}{2 D}\left[u_{i k}^{Y}(x, y+D, z)-u_{i k}^{Y}(x, y-D, z)\right]  \tag{A.6}\\
& \frac{\partial u_{i k}^{Y}}{\partial z} \approx \frac{1}{2 D}\left[u_{i k}^{Y}(x, y, z+D)-u_{i k}^{Y}(x, y, z-D)\right] \tag{A.7}
\end{align*}
$$

where $x, y, z$ are the coordinates of the source point $p$ in the global coordinate system $O x y z$ and $D$ is the distance between the two source points. The derivatives of the tractions in Eq. (A.4) can also be calculated using the above methods.

In order to calculate the derivatives $u_{i k, j}^{Y}$ and $t_{i k, j}^{Y}$, it is necessary to compute the displacements and stresses of the fundamental solution at six points in the neighborhood of the source point $p$. The choice of the interval $D$ is a crucial decision and an extensive numerical investigation is executed in Pan and Yuan [35] and Tonon et al. [38]. The best $D$ value is $10^{-6} r$ where $r$ is the distance between the source and field points.

## Appendix B. The fundamental solution in $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ coordinates

For a layered medium, shown in Fig. 3, the vector fields of displacements, vertical stresses and plane strains are defined as

$$
\boldsymbol{u}=\left[\begin{array}{lll}
u_{x^{\prime}} & u_{y^{\prime}} & u_{z^{\prime}}
\end{array}\right]^{\mathrm{T}}, \quad \boldsymbol{T}_{z^{\prime}}=\left[\begin{array}{lll}
\sigma_{x^{\prime} z^{\prime}} & \sigma_{y^{\prime} z^{\prime}} & \sigma_{z z^{\prime}}
\end{array}\right]^{\mathrm{T}}, \quad \boldsymbol{\Gamma}_{p}=\left[\begin{array}{lll}
\varepsilon_{x^{\prime} x^{\prime}} & \varepsilon_{x^{\prime} y^{\prime}} & \varepsilon_{y^{\prime} y^{\prime}} \tag{B.1}
\end{array}\right]^{\mathrm{T}}
$$

where the superscript T stands for the transpose of a matrix.
For concentrated point loads, i.e., $\boldsymbol{F}_{c}=\left[\begin{array}{lll}f_{x^{\prime}} & f_{y^{\prime}} & f_{z^{\prime}}\end{array}\right]^{\mathrm{T}}$, the solution expressions of the layer medium are presented as follows

$$
\begin{equation*}
\boldsymbol{u}=\boldsymbol{G}_{u} \boldsymbol{F}_{c}, \quad \boldsymbol{T}_{z}=\boldsymbol{G}_{z} \boldsymbol{F}_{c}, \quad \boldsymbol{\Gamma}_{z}=\boldsymbol{G}_{p} \boldsymbol{F}_{c} . \tag{B.2}
\end{equation*}
$$

where the matrices $\boldsymbol{G}_{u}, \boldsymbol{G}_{z}$ and $\boldsymbol{G}_{p}$ can be expressed as

$$
\begin{align*}
& 2 \pi \boldsymbol{G}_{u}\left(x^{\prime}, y^{\prime}, z^{\prime}\right)=\int_{0}^{\infty}\left(\begin{array}{ccc}
\Phi_{1} J_{0}-\frac{x^{\prime 2}-y^{\prime 2}}{r^{2}} \Phi_{2} J_{2} & -\frac{2 x^{\prime} y^{\prime}}{r^{2}} \Phi_{2} J_{2} & -\frac{x^{\prime}}{r} \Phi_{13} J_{1} \\
-\frac{2 x^{\prime} y^{\prime}}{r^{2}} \Phi_{2} J_{2} & \Phi_{1} J_{0}+\frac{x^{\prime 2}-y^{\prime 2}}{r^{2}} \Phi_{2} J_{2} & -\frac{y^{\prime}}{r} \Phi_{13} J_{1} \\
\frac{x^{\prime}}{r} \Phi_{31} J_{1} & \frac{y^{\prime}}{r} \Phi_{31} J_{1} & \Phi_{33} J_{0}
\end{array}\right) d \rho  \tag{B.3}\\
& 2 \pi \boldsymbol{G}_{z}\left(x^{\prime}, y^{\prime}, z^{\prime}\right)=\int_{0}^{\infty}\left(\begin{array}{ccc}
\Psi_{1} J_{0}-\frac{x^{\prime 2}-y^{\prime 2}}{r^{2}} \Psi_{2} J_{2} & -\frac{2 x^{\prime} y^{\prime}}{r^{2}} \Psi_{2} J_{2} & -\frac{x^{\prime}}{r} \Psi_{13} J_{1} \\
-\frac{2 x^{\prime} y^{\prime}}{r^{2}} \Psi_{2} J_{2} & \Psi_{1} J_{0}+\frac{x^{\prime 2}-y^{\prime 2}}{r^{2}} \Psi_{2} J_{2} & -\frac{y^{\prime}}{r} \Psi_{13} J_{1} \\
\frac{x^{\prime}}{r} \Psi_{31} J_{1} & \frac{y^{\prime}}{r} \Psi_{31} J_{1} & \Psi_{33} J_{0}
\end{array}\right) \rho d \rho \tag{B.4}
\end{align*}
$$

$$
\begin{align*}
2 \pi \boldsymbol{G}_{p}\left(x^{\prime}, y^{\prime}, z^{\prime}\right)= & -\frac{1}{2} \int_{0}^{\infty}\left(\begin{array}{ccc}
\frac{x^{\prime}}{r}\left(2 \Phi_{1}+\Phi_{2}\right) J_{1} & \frac{y^{\prime}}{r} \Phi_{2} J_{1} & \Phi_{13} J_{0} \\
\frac{y^{\prime}}{r} \Phi_{1} J_{1} & \frac{x^{\prime}}{r} \Phi_{1} J_{1} & 0 \\
\frac{x^{\prime}}{r} \Phi_{2} J_{1} & \frac{y^{\prime}}{r}\left(2 \Phi_{1}+\Phi_{2}\right) J_{1} & \Phi_{13} J_{0}
\end{array}\right) \rho d \rho \\
+ & \left(\begin{array}{lll}
\left(\frac{4 x^{\prime 3}}{r^{3}}-\frac{3 x^{\prime}}{r}\right) \int_{0}^{\infty} J_{3} & \left(\frac{3 y^{\prime}}{r}-\frac{4 y^{\prime 3}}{r^{3}}\right) \Phi_{2} J_{3} & \frac{x^{\prime 2}-y^{\prime 2}}{r^{2}} \Phi_{13} J_{2} \\
\left(\frac{3 y^{\prime}}{r}-\frac{4 y^{\prime 3}}{r^{3}}\right) \Phi_{2} J_{3} & \left(\frac{3 x^{\prime}}{r}-\frac{4 x^{\prime 3}}{r^{3}}\right) \Phi_{2} J_{3} & \frac{2 x^{\prime} y^{\prime}}{r^{2}} \Phi_{13} J_{2} \\
\left(\frac{3 x^{\prime}}{r}-\frac{4 x^{\prime 3}}{r^{3}}\right) \Phi_{2} J_{3} & \left(\frac{4 y^{\prime 3}}{r^{3}}-\frac{3 y^{\prime}}{r}\right) \Phi_{2} J_{3} & \frac{y^{\prime 2}-x^{\prime 2}}{r^{2}} \Phi_{13} J_{2}
\end{array}\right) \rho d \rho \tag{B.5}
\end{align*}
$$

where $\quad r=\sqrt{x^{\prime 2}+y^{\prime 2}}, \quad \Phi_{1}=\frac{1}{2}\left[\Phi_{11}+\Phi_{22}\right], \quad \Phi_{2}=\frac{1}{2}\left[\Phi_{11}-\Phi_{22}\right], \quad \Psi_{1}=\frac{1}{2}\left[\Psi_{11}+\Psi_{22}\right] \quad$ and $\Psi_{2}=\frac{1}{2}\left[\Psi_{11}-\Psi_{22}\right]$. The following identity of Bessel functions ( $J_{m}$ ) of order $m$ is used in the further reduction
$\frac{1}{2 \pi} \int_{0}^{2 \pi} e^{ \pm i(\rho r \sin \psi-m \psi)} d \psi=J_{m}(\rho r)=J_{m}, \quad m=0, \pm 1, \pm 2, \cdots$.
The expressions of the matrices $\boldsymbol{G}_{u}, \boldsymbol{G}_{z}$ and $\boldsymbol{G}_{p}$ include the inverse Hankel transform integrals with a semi-infinite interval. The expressions also include the Bessel functions of order of zero, unit and second and ten fundamental functions of $\Phi_{11}, \Phi_{13}, \Phi_{22}, \Phi_{31}, \Phi_{33}$, $\Psi_{11}, \Psi_{13}, \Psi_{22}, \Psi_{31}$, and $\Psi_{33}$. A proceeding limit technique, based on an adaptively iterative Simpson's quadrature, is adopted in the evaluation of the inverse Hankel transform integrals in Eqs. (B.3) to (B.5). Using the procedure, the semi-infinite interval of the inverse Hankel transform integrals can be accommodated and the improper integrals can be efficiently evaluated with high and controlled accuracy.

The solutions of the plane stresses $\left(\sigma_{x^{\prime} x^{\prime}}, \sigma_{x^{\prime} y^{\prime}}, \sigma_{y^{\prime} y^{\prime}}\right)$ and the vertical strains $\left(\varepsilon_{x z^{\prime}}, \varepsilon_{y z^{\prime}}, \varepsilon_{z z^{\prime}}\right)$ due to the point loads $\left(f_{x^{\prime}}, f_{y^{\prime}}, f_{z^{\prime}}\right)$ can be easily obtained from the solutions of the vertical stresses $\left(\sigma_{x^{\prime} \prime}^{\prime}, \sigma_{y^{\prime} z^{\prime}}, \sigma_{z^{\prime} z^{\prime}}\right)$ and the plane strains $\left(\varepsilon_{x^{\prime} x^{\prime}}, \varepsilon_{x^{\prime} y^{\prime}}, \varepsilon_{y^{\prime} y^{\prime}}\right)$ by using the constitutive equations.

More details of the expressions and mathematical properties of the fundamental solution can be found in Yue [12].

## Appendix C. The fundamental solution in Oxyz coordinates

C.1.The solution in $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ coordinates due to $f_{x}$

The point force $f_{x}$ in $O x y z$ can be re-expressed as follows in the coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$,

$$
\begin{equation*}
f_{x^{\prime}}=f_{x} \cos \theta, \quad f_{y^{\prime}}=0, \quad f_{z^{\prime}}=-f_{x} \sin \theta \tag{C.1}
\end{equation*}
$$

The displacements and stresses in the system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ induced by $f_{x}=1$ can be calculated using the equations
$u_{i^{\prime}}=\left.u_{i^{\prime}}\right|_{f_{x}=1} \cos \theta-\left.u_{i^{\prime}}\right|_{f_{z}=1} \sin \theta$
$\sigma_{i j^{\prime}}=\left.\sigma_{i j^{\prime}}\right|_{f_{x}=1} \cos \theta-\left.\sigma_{i j^{\prime}}\right|_{f_{z}=1} \sin \theta$
where $\left.u_{i^{\prime}}\right|_{f_{x}=1}$ and $\left.\sigma_{i j^{\prime}}\right|_{f_{x}=1}$ are the displacements and stresses induced by $f_{x^{\prime}}=1$, and $\left.u_{i i^{\prime}}\right|_{f_{z}=1}$ and $\left.\sigma_{i j^{\prime}}\right|_{f_{z}=1}$ are the displacements and stresses induced by $f_{z^{\prime}}=1$, respectively.

## C.2. The solution in $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ coordinates due to $f_{y}$

The point force $f_{y}$ in Oxyz can be re-expressed as follows in the coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$,
$f_{x^{\prime}}=0, \quad f_{y^{\prime}}=f_{y}, \quad f_{z^{\prime}}=0$
The displacements and stresses in the $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ system induced by $f_{y}=1$ can be calculated using the equations

$$
\begin{align*}
& u_{i^{\prime}}=\left.u_{i^{\prime}}\right|_{f_{y}=1}  \tag{C.5}\\
& \sigma_{i j^{\prime}}=\left.\sigma_{i j^{\prime}}\right|_{f_{y}=1} \tag{C.6}
\end{align*}
$$

where $\left.u_{i^{\prime}}\right|_{f_{y}=1}$ and $\left.\sigma_{i j^{\prime}}\right|_{f_{y}=1}$ are the displacements and stresses induced by $f_{y^{\prime}}=1$.

## C.3. The solution in $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ coordinates due to $f_{z}$

The point force $f_{z}$ in $O x y z$ can be re-expressed as follows in the coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$,
$f_{x^{\prime}}=f_{z} \sin \theta, \quad f_{y^{\prime}}=0, \quad f_{z^{\prime}}=f_{z} \cos \theta$
The displacements and stresses in the system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ induced by $f_{z}=1$ can be calculated using the equations
$u_{i^{\prime}}=\left.u_{i^{\prime}}\right|_{f_{x}=1} \sin \theta+\left.u_{i^{\prime}}\right|_{f_{z^{\prime}}=1} \cos \theta$
$\sigma_{i j^{\prime}}=\left.\sigma_{i j^{\prime}}\right|_{f_{x}=1} \sin \theta+\left.\sigma_{i j^{\prime}}\right|_{f_{z}=1} \cos \theta$
where $\left.u_{i^{\prime}}\right|_{f_{x}=1},\left.\sigma_{i j^{\prime}}\right|_{f_{x}=1},\left.u_{i^{\prime}}\right|_{f_{z^{\prime}}=1}$ and $\left.\sigma_{i j^{\prime}}\right|_{f_{z}=1}$ have the same meanings as defined in Eqs. (C.2) and (C.3).

## C.4. The solutions in Oxyz coordinates

The generalized Kelvin solution ( $u_{i}$ and $\sigma_{i j}$ ) in Oxyz coordinates induced by the point body force vector $\left(f_{x}, f_{y}, f_{z}\right)$ have the following matrix transformation relationship with the generalized Kelvin in $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ coordinates induced by the point body force vector $\left(f_{x^{\prime}}, f_{y^{\prime}}, f_{z^{\prime}}\right)$
$u_{i}=\beta_{i i i^{\prime}} u_{i^{\prime}} \quad\left(i=x, y, z, i^{\prime}=x^{\prime}, y^{\prime}, z^{\prime}\right)$
$\sigma_{i j}=\beta_{i i^{\prime}} \beta_{j j^{\prime}} \sigma_{i j^{\prime}} \quad\left(i, j=x, y, z, i^{\prime}, j^{\prime}=x^{\prime}, y^{\prime}, z^{\prime}\right)$
where $\beta_{i i^{\prime}}$ are the coefficients of the coordinate transform matrix.

The coordinates at any point in the coordinate system $O^{\prime} x^{\prime} y^{\prime} z^{\prime}$ can be obtained from the ones in the coordinate system Oxyz by

$$
\begin{align*}
& x^{\prime}=\beta_{x^{\prime} x}\left(x-d_{1}\right)+\beta_{x^{\prime} y} y+\beta_{x^{\prime} z} z  \tag{C.12}\\
& y^{\prime}=\beta_{y^{\prime} x}\left(x-d_{1}\right)+\beta_{y^{\prime} y} y+\beta_{z^{\prime} z} z \tag{C.13}
\end{align*}
$$

$z^{\prime}=\beta_{z^{\prime} x}\left(x-d_{1}\right)+\beta_{z^{\prime} y} y+\beta_{z^{\prime} z} z$
where $d_{1}$ is the distance between the two origins $O$ and $O^{\prime}$, and $\beta_{i^{\prime} i}$ are the coefficients of the coordinate transform.

## Appendix D. The interpolation functions for five element types

D.1. Shape functions for continuous finite boundary element (type I)
$N_{1}=-0.25(1-\xi)(1-\eta)(1+\xi+\eta)$
$N_{2}=-0.25(1+\xi)(1-\eta)(1-\xi+\eta)$
$N_{3}=-0.25(1+\xi)(1+\eta)(1-\xi-\eta)$
$N_{4}=-0.25(1-\xi)(1+\eta)(1+\xi-\eta)$
$N_{5}=0.5\left(1-\xi^{2}\right)(1-\eta)$
$N_{6}=0.5\left(1-\eta^{2}\right)(1+\xi)$
$N_{7}=0.5\left(1-\xi^{2}\right)(1+\eta)$
$N_{8}=0.5\left(1-\eta^{2}\right)(1-\xi)$
D.2. Shape functions for discontinuous finite boundary element (type II)
$N_{1}=-0.3(1-\xi)(1-\eta)(1+\xi+1.5 \eta)$
$N_{2}=-0.3(1+\xi)(1-\eta)(1-\xi+1.5 \eta)$
$N_{3}=-0.3(1+\xi)(2 / 3+\eta)(1-\xi-\eta)$
$N_{4}=-0.3(1-\xi)(2 / 3+\eta)(1+\xi-\eta)$
$N_{5}=0.6\left(1-\xi^{2}\right)(1-\eta)$
$N_{6}=0.75(1+\xi)(1-\eta)(2 / 3+\eta)$
$N_{7}=0.6\left(1-\xi^{2}\right)(2 / 3+\eta)$
$N_{8}=0.75(1-\xi)(1-\eta)(2 / 3+\eta)$
D.3. Interpolation functions for continuous infinite elements (type III)
D.3.1. Shape functions

$$
\begin{align*}
& { }^{\infty} N_{1}=\xi \eta(1-\xi) /(1-\eta) \\
& { }^{\infty} N_{2}=-\xi \eta(1+\xi) /(1-\eta) \\
& { }^{\infty} N_{3}=0.5 \xi(1+\xi)(1+\eta) /(1-\eta) \\
& { }^{\infty} N_{4}=0.5 \xi(\xi-1)(1+\eta) /(1-\eta) \\
& { }^{\infty} N_{5}=-2 \eta\left(1-\xi^{2}\right) /(1-\eta) \\
& { }^{\infty} N_{6}=\left(1-\xi^{2}\right)(1+\eta) /(1-\eta) \tag{D.3}
\end{align*}
$$

D.3.2. Interpolation functions of displacements

$$
\begin{align*}
& { }^{\infty} N_{1}^{u}=0.25 \xi(\xi-1)(1-\eta) \\
& { }^{\infty} N_{2}^{u}=0.25 \xi(\xi+1)(1-\eta) \\
& { }^{\infty} N_{5}^{u}=0.5\left(1-\xi^{2}\right)(1-\eta) \tag{D.4}
\end{align*}
$$

D.3.3. Interpolation functions of tractions

$$
\begin{align*}
& { }^{\infty} N_{1}^{t}=0.125 \xi(\xi-1)(1-\eta)^{2} \\
& { }^{\infty} N_{2}^{t}=0.125 \xi(\xi+1)(1-\eta)^{2} \\
& { }^{\infty} N_{5}^{t}=0.25\left(1-\xi^{2}\right)(1-\eta)^{2} \tag{D.5}
\end{align*}
$$

D.4. Interpolation functions for discontinuous infinite elements (type IV)

## D.4.1. Shape functions

$$
\begin{aligned}
& { }^{\infty} N_{1}=-1.8 \xi \eta(\xi-1) /(1-\eta) \\
& { }^{\infty} N_{2}=-1.2 \xi \eta(\xi+2 / 3) /(1-\eta) \\
& { }^{\infty} N_{3}=0.6 \xi(\xi+2 / 3)(1+\eta) /(1-\eta) \\
& { }^{\infty} N_{4}=0.9 \xi(\xi-1)(1+\eta) /(1-\eta) \\
& { }^{\infty} N_{5}=-3 \eta(1-\xi)(2 / 3+\xi) /(1-\eta)
\end{aligned}
$$

$$
\begin{equation*}
{ }^{\infty} N_{6}=1.5(1-\xi)(2 / 3+\xi)(1+\eta) /(1-\eta) \tag{D.6}
\end{equation*}
$$

D.4.2. Interpolation functions of displacements

$$
\begin{align*}
& { }^{\infty} N_{1}^{u}=0.45(1-\eta) \xi(\xi-1) \\
& { }^{\infty} N_{2}^{u}=0.3(1-\eta) \xi(\xi+2 / 3) \\
& { }^{\infty} N_{5}^{u}=0.75(1-\eta)(1-\xi)(\xi+2 / 3) \tag{D.7}
\end{align*}
$$

D.4.3. Interpolation functions of tractions

$$
\begin{align*}
& { }^{\infty} N_{1}^{t}=0.225(1-\eta)^{2} \xi(\xi-1) \\
& { }^{\infty} N_{2}^{t}=0.15(1-\eta)^{2} \xi(2 / 3+\xi) \\
& { }^{\infty} N_{5}^{t}=0.375(1-\eta)^{2}(1-\xi)(2 / 3+\xi) \tag{D.8}
\end{align*}
$$

D.5. Interpolation functions for discontinuous infinite elements (type $V$ )
D.5.1. Shape functions

$$
\begin{align*}
& { }^{\infty} N_{1}=-1.2 \eta(\xi-2 / 3) /(1-\eta) \\
& { }^{\infty} N_{2}=-1.8 \xi \eta(\xi+1) /(1-\eta) \\
& { }^{\infty} N_{3}=0.9 \xi(\xi+1)(1+\eta) /(1-\eta) \\
& { }^{\infty} N_{4}=0.6 \xi(\xi-2 / 3)(1+\eta) /(1-\eta) \\
& { }^{\infty} N_{5}=-3 \eta(2 / 3-\xi)(1+\xi) /(1-\eta) \\
& { }^{\infty} N_{6}=1.5(2 / 3-\xi)(1+\xi)(1+\eta) /(1-\eta) \tag{D.9}
\end{align*}
$$

D.5.2. Interpolation functions of displacements

$$
\begin{align*}
& { }^{\infty} N_{1}^{u}=0.3(1-\eta) \xi(\xi-2 / 3) \\
& { }^{\infty} N_{2}^{u}=0.45(1-\eta) \xi(\xi+1) \\
& { }^{\infty} N_{5}^{u}=0.75(1-\eta)(2 / 3-\xi)(1+\xi) \tag{D.10}
\end{align*}
$$

D.5.3. Interpolation functions of tractions

$$
{ }^{\infty} N_{1}^{t}=0.15(1-\eta)^{2} \xi(\xi-2 / 3)
$$

$$
\begin{align*}
& { }^{\infty} N_{2}^{t}=0.225(1-\eta)^{2} \xi(\xi+1) \\
& { }^{\infty} N_{5}^{t}=0.375(1-\eta)^{2}(2 / 3-\xi)(1+\xi) \tag{D.11}
\end{align*}
$$

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Fig. 1. A layered halfspace subject to distributed loadings on the boundary surface


Fig. 2. Definition of $S_{F}$ and $S_{I}$ on the boundary surface of a non-horizontally layered halfspace


Fig. 3. A multilayered elastic solid of infinite extent subjected to the body forces $f_{x^{\prime}}, f_{y^{\prime}}$ and $f_{z^{\prime}}$ concentrated at a point $(0,0, d)$


Fig. 4. Relation between two coordinate systems

(5a) continuous element (type I)

(5b) discontinuous element (type II)
Fig. 5. Two types of finite boundary elements

(6a) continuous element (type III)

(6b) discontinuous element (type IV)

(6c) discontinuous element (type V)
Fig. 6. Three types of infinite boundary elements

(7a) node 1

(7b) node 2

(7c) node 5
Fig. 7. The triangle domain divided in a continuous infinite element (type IV) for different nodes


Fig. 8. Bi-material elastic halfspace with an inclined interface plane and subjected to normal pressure over a square area on the horizontal boundary surface ( $\boldsymbol{v}_{\mathbf{1}}=\boldsymbol{v}_{2}=\mathbf{0}$ )


Fig. 9. The boundary mesh on the boundary surface with 1192 nodes for 284 finite continuous elements (I), 36 finite discontinuous elements (II), 60 infinite continuous elements (III) and 4 infinite discontinuous elements (2 for IV and 2 for V)

(10a) horizontal boundary surface

(10b) vertical cross-section
Fig. 10. A square footing acting on the strike line area of a bi-material elastic halfspace with an arbitrarily inclined interface plane

(11a) displacement $u_{x}$

(11b) displacement $u_{y}$

(11c) displacement $u_{z}$
Fig. 11. Variation of three displacement components ( $y=0.5 \mathrm{~m}$ and $\mathrm{z}=1 \mathrm{~m}$ )

(12a) stress $\sigma_{x x}$

(12b) stress $\sigma_{x y}$

(12c) stress $\sigma_{x z}$

(12d) stress $\sigma_{y y}$

(12e) stress $\sigma_{y z}$

(12f) stress $\sigma_{z z}$
Fig. 12. Variation of stress components ( $y=0.5 \mathrm{~m}$ and $\mathrm{z}=1 \mathrm{~m}$ )

(13a) displacement $u_{x}$

(13b) displacement $u_{y}$

(13c) displacement $u_{z}$
Fig.13. Variation of displacements with the horizontal distance $y$

(14a) stress $\sigma_{x x}$

(14b) stress $\sigma_{x y}$


(14d) stress $\sigma_{y y}$


Fig. 14. Variation of stresses with the horizontal distance $y\left(\theta=60^{\circ}, z=1 \mathrm{~m}\right)$

(15a) displacement $u_{x}$

(15b) displacement $u_{y}$

(15c) displacement $u_{z}$
Fig. 15. Variation of displacements with the depth $z\left(\theta=60^{\circ}, y=0.5 \mathrm{~m}\right)$


Fig. 16. Variation of Mises stress with the depth $z\left(\theta=60^{\circ}, y=0.5 \mathrm{~m}\right)$

(17a) horizontal boundary surface

(17b) vertical cross-section
Fig. 17. Square footing on three-material halfspace with two non-horizontally placed and parallel interfaces

(18a) displacement $u_{x}$

(18b) displacement $u_{y}$

(18c) displacement $u_{z}$
Fig. 18. Variation of three displacement components with the angle $\theta(y=0.5 \mathrm{~m}$ and $\mathrm{z}=1 \mathrm{~m})$

(19a) stress $\sigma_{x x}$

(19b) stress $\sigma_{x y}$

(19c) stress $\sigma_{x z}$

(19d) stress $\sigma_{y y}$

(19e) stress $\sigma_{y z}$

(19f) stress $\sigma_{z z}$
Fig. 19. Variation of six stress components with the angle $\theta$ ( $y=0.5 \mathrm{~m}$ and $z=1 \mathrm{~m}$ )

Table 1. Stress jumps $\Delta \sigma_{i j}(\mathrm{MPa})$ across the material interface at $z=1 \mathrm{~m}$ and $y=0.5 \mathrm{~m}$ for different $\theta$
in the bi-material and three-material halfspaces

| $\begin{aligned} & \hline \text { Stress } \\ & \text { jump } \end{aligned}$ | $\theta\left({ }^{\circ}\right)$ | bi-material halfspace | three-material halfspace |  |
| :---: | :---: | :---: | :---: | :---: |
|  |  | $x=c^{\prime}$ | $x=c_{1}^{\prime}$ | $x=c_{2}^{\prime}$ |
| $\Delta \sigma_{x x}$ | 30 | 7.1754 | 8.3462 | 2.6803 |
|  | 60 | 7.3931 | 4.4729 | 3.9206 |
|  | 90 | 0.0009 | 0.0007 | 0.0006 |
|  | 120 | 9.0623 | 3.3704 | 4.4491 |
|  | 150 | 11.2755 | 2.1995 | 8.2961 |
| $\Delta \sigma_{x y}$ | 30 | 1.2214 | 5.8617 | 0.6966 |
|  | 60 | 3.1317 | 3.6639 | 1.2041 |
|  | 90 | 0.0002 | 0.0001 | 0.0009 |
|  | 120 | 4.2760 | 1.2354 | 3.6809 |
|  | 150 | 2.6902 | 0.8616 | 5.8195 |
| $\Delta \sigma_{x z}$ | 30 | 4.1517 | 4.9881 | 1.7107 |
|  | 60 | 12.7937 | 7.7645 | 5.9670 |
|  | 90 | 0.0005 | 0.0002 | 0.0003 |
|  | 120 | 15.6937 | 6.5093 | 7.7229 |
|  | 150 | 6.5013 | 1.9526 | 4.9945 |
| $\Delta \sigma_{y y}$ | 30 | 1.3539 | 8.8661 | 1.3316 |
|  | 60 | 0.1489 | 6.6653 | 2.1390 |
|  | 90 | 0.1862 | 0.3373 | 0.3216 |
|  | 120 | 1.8660 | 2.5628 | 6.6584 |
|  | 150 | 0.4631 | 0.9038 | 8.7963 |
| $\Delta \sigma_{y z}$ | 30 | 0.7079 | 3.4214 | 0.5584 |
|  | 60 | 5.4388 | 6.3572 | 2.4775 |
|  | 90 | 9.0441 | 5.4129 | 5.4097 |
|  | 120 | 7.3972 | 1.9396 | 6.3861 |
|  | 150 | 1.5502 | 0.9062 | 3.4983 |
| $\Delta \sigma_{z z}$ | 30 | 2.4086 | 2.5032 | 0.8887 |
|  | 60 | 22.2426 | 13.4394 | 10.9973 |
|  | 90 | 41.1799 | 22.0673 | 22.0424 |
|  | 120 | 27.1359 | 10.9521 | 13.3676 |
|  | 150 | 3.7430 | 0.2147 | 2.3860 |


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